

ArduPilot Conference 2022 Tailsitter update

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What is a tailsitter?

A VTOL plane that hovers at a different attitude to that it uses in forward flight.

Convair XFY Pogo – experimental tailsitter 1954



ARDUP LOT Versatile, Trusted, Open

Types



Dual motor - Marco Robustini's Explora



Single motor – Tridge's Addiction X



Vectored dual motor- FT Bronco



Copter - SwanK1 – off the shelf AP compatible





Copter – Pierre Losa's Jetwing

Contents



- Smaller changes
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- Transitions
- Weathervaning



https://youtu.be/5W3f13zCWsM

Smaller changes



- **Q_TAILSIT_ENABLE** #18015
 - Hides params for none tailsitter users
 - Removes odd combinations of Q_FRAME_CLASS and Q_TAILSIT_MOTMX to enable tailsitter
- Set level in both refrence frames #18311
 - FW mode calibrate roll and pitch
 - VTOL mode calibrate pitch and yaw
- Transition to and from inverted FW flight #17118
- Dedicated build for SwanK1 #18254
- EKF3 nose up init issue fixed #15351
 - No more YAW inconsistent warnings

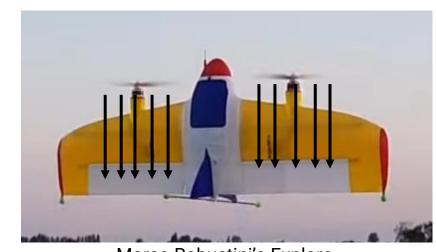


Control surface gain scailing

Primarily used on dual motor none vectored. Control surfaces are only method for actuation of pitch and yaw.

Response is dictated by airflow over the control surfaces, this depends on thrust and forward speed.

New option estimates the airspeed seen by control surfaces.



Marco Robustini's Explora

Q_TAILSIT_GSCMSK = 4 and **Q_TAILSIT_DSKLD** > 0

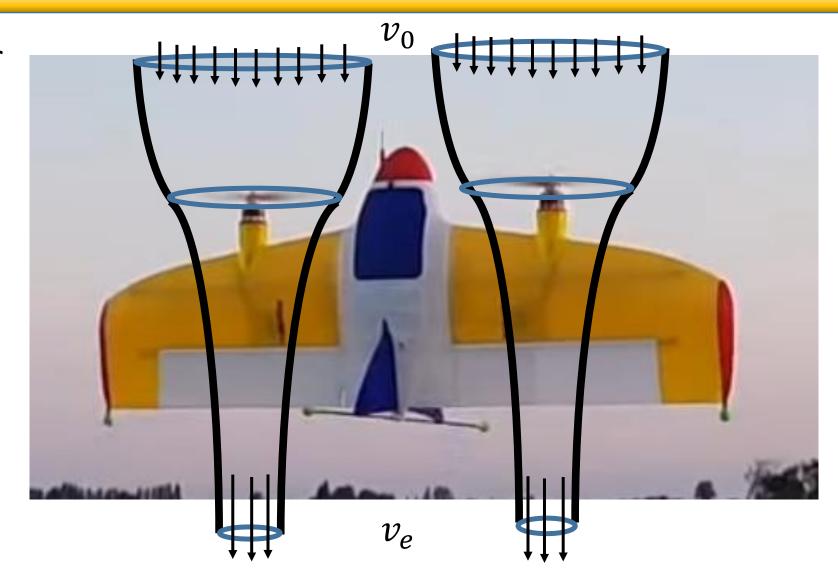


Conservation of mass over control volume.

Instantons pressure change over disk.

<u>Derived</u> from Bernoulli's equation.

$$T = \rho A_{disk} \frac{v_e^2 - v_0^2}{2}$$





1.
$$T = \rho A_{disk} \frac{v_e^2 - v_0^2}{2}$$

5.
$$v_e^2 = \frac{m}{A_{disk}} \frac{\frac{\iota}{t_{hover}}^2 2g}{\rho} + v_0^2$$

2.
$$v_e^2 = \frac{2T}{\rho A_{disk}} + v_0^2$$

6. Disk loading =
$$\frac{m}{A_{disk}}$$

3.
$$T = \frac{t}{t_{hover}} mg$$

Disk loading, set with **Q_TAILSIT_DSKLD**. Vehicle mass divided by total disk area of propellers.

4.
$$v_e^2 = \frac{\frac{\iota}{t_{hover}} 2mg}{\rho A_{disk}} + v_0^2$$



Normalise by hover to give scaling of 1 in hover case, does not mess up PIDs

Hover case:
$$\frac{t}{t_{hover}} = 1$$
, $v_0 = 0$

General case: v_0 = airspeed, from sensor or EKF

Control surface gain:
$$\frac{v_{e\,hover}^2}{v_e^2} = \frac{\frac{\frac{m}{A_{disk}\,\rho}}{\frac{t}{A_{disk}}\,\rho}}{\frac{m}{A_{disk}}\frac{\frac{t}{t_{hover}}^2g}{\rho} + v_0^2}$$

Scaling constrained between **Q_TAILSIT_GSCMIN** and **Q_TAILSIT_GSCMAX**

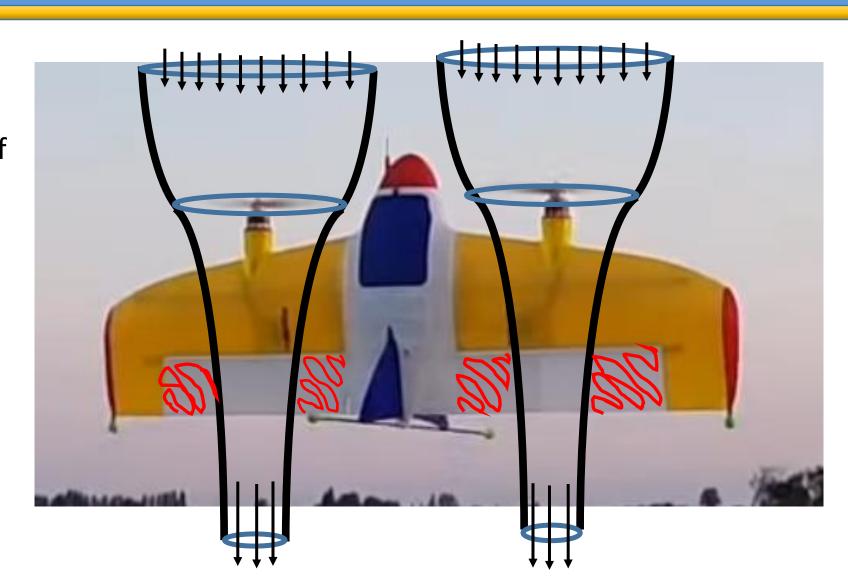


Control volume does not cover all of the control surface.

To compensate a lower value of **Q_TAILSIT_DSKLD** is used to represent the average airflow over the control surface. Half area gives half **Q_TAILSIT_DSKLD**.

Tune by using high throttle to gain airspeed and reducing to low throttle and coasting. This covers a range of throttles and airspeeds.

Lower **Q_TAILSIT_DSKLD** if oscillations at high throttle.



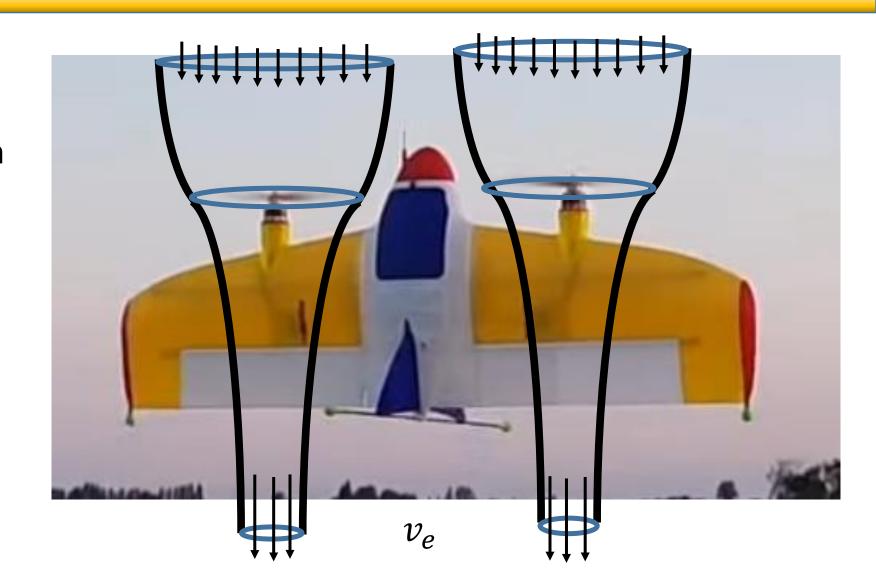
Min outflow speed - #19249



Q_TAILSIT_MIN_VO

Requires disk theory gain scailing to be setup.

Increases throttle to always give minimum airflow over control surfaces.



Min outflow speed - #19249



1.
$$T = \rho A_{disk} \frac{v_e^2 - v_0^2}{2}$$

2.
$$T = \frac{t}{t_{hover}} mg$$

3.
$$t = \frac{\rho A_{disk} \frac{v_e^2 - v_0^2}{2}}{mg} t_{hover}$$

4.
$$v_e = q_{\text{TAILSIT_MIN_VO}}$$

Climb case, $v_0 = airspeed$

$$t = \frac{\rho A_{disk} \frac{v_e^2 - v_0^2}{2}}{mg} t_{hover}$$

Climb case, $v_0 = \text{body frame X velocity}$

$$t = \frac{\rho A_{disk} \frac{v_e^2 + v_0^2}{2}}{mg} t_{hover}$$

Min outflow speed - #19249



Results in low minimum throttle limit with forward airspeed, higher minimum throttle with no or negative airspeed.

Q_TAILSIT_MIN_VO too high will result in flyaway.

Need to tune, not all of control surface in stream tube. Increase until minimum throttle does not loose control when descending.

Maximum throttle limit constrained to value derived from **Q_A_THR_MIX_MAN.** High values result in larger posible increase in throttle.

Control surface less - #19249



Bad idea - With airspeed prop effectiveness decreases and aerodynamic

forces increase. Fly too fast -> crash!

Q_TAILSIT_ENABLE = 2

Enables active stabilization in forward flight.

Forces "air-mode" - stabilization active at zero throttle in forward flight.

See **Q_A_THR_MIX_*** params again.

Control surfaces with motors - #14453, #19463



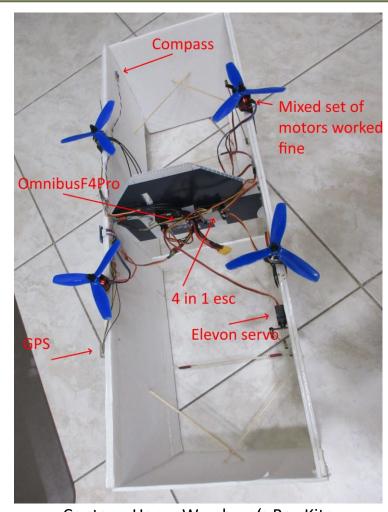
Much better, no problems at high speed.

Q_TAILSIT_VT_R_P, Q_TAILSIT_VT_P_P and Q_TAILSIT_VT_Y_P scale control surfaces relative to motors.

Q_ASSIST_SPEED to enable and disable motors based on airspeed in forward flight.

Q_OPTIONS: 8 – Motors only q assist - use plane control surface outputs and I terms.

Q_TAILSIT_MOTMX: to select active motors in FW, 15 here

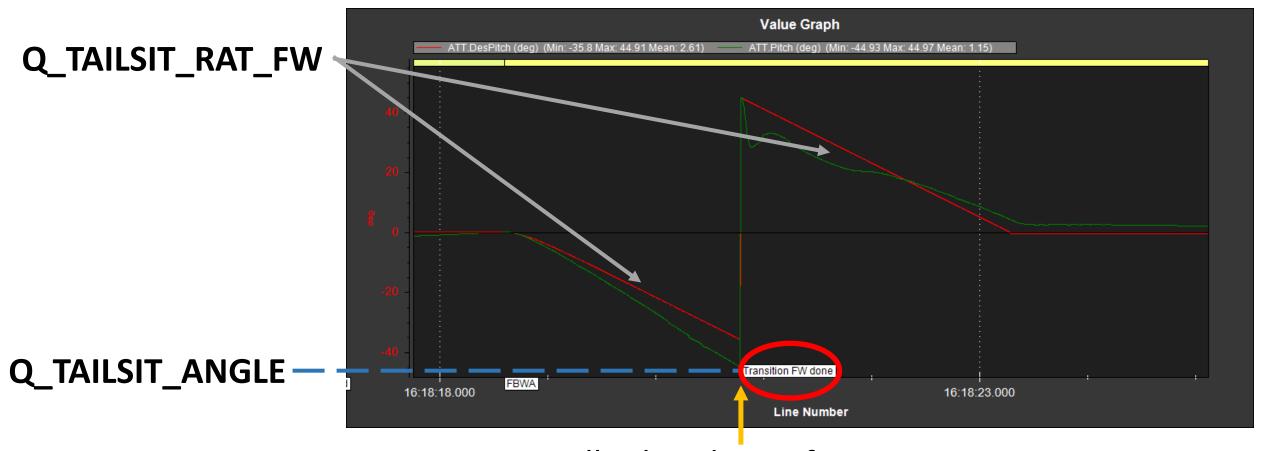


Copter – Henry Wurzburg's Box Kite

Transitions – to Forward flight



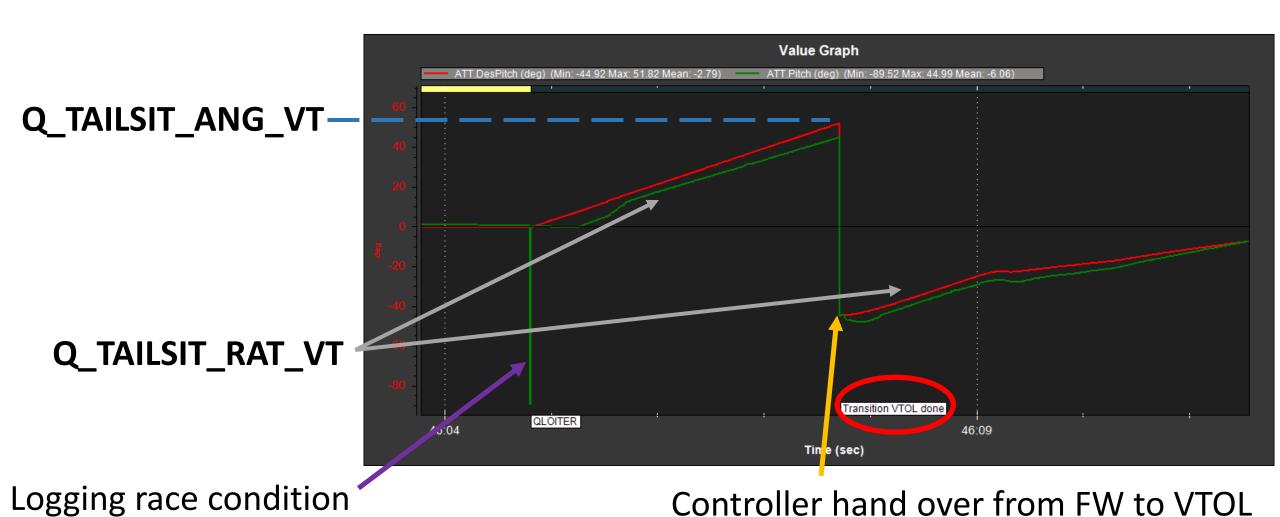
Too many improments to mention! Must have good VTOL and FW tune first.



Controller hand over from VTOL to FW

Transitions – to VTOL flight





Transitions – to VTOL flight



Q_TAILSIT_MIN_VO

Q_A_THR_MIX_MIN
Q_A_THR_MIX_MAX
Q_A_THR_MIX_MAN

To bump throttle to min level.

Q_TAILSIT_THR_VT



Weathervaning - #19859



Q_WVANE_ENABLE
Q_WVANE_TAKEOFF
Q_WVANE_LAND
Nose into wind,

Nose or tail into wind,

Side into wind, tail into wind



Q_WVANE_GAIN

https://youtu.be/SS7y0tQfWjg

Questions?



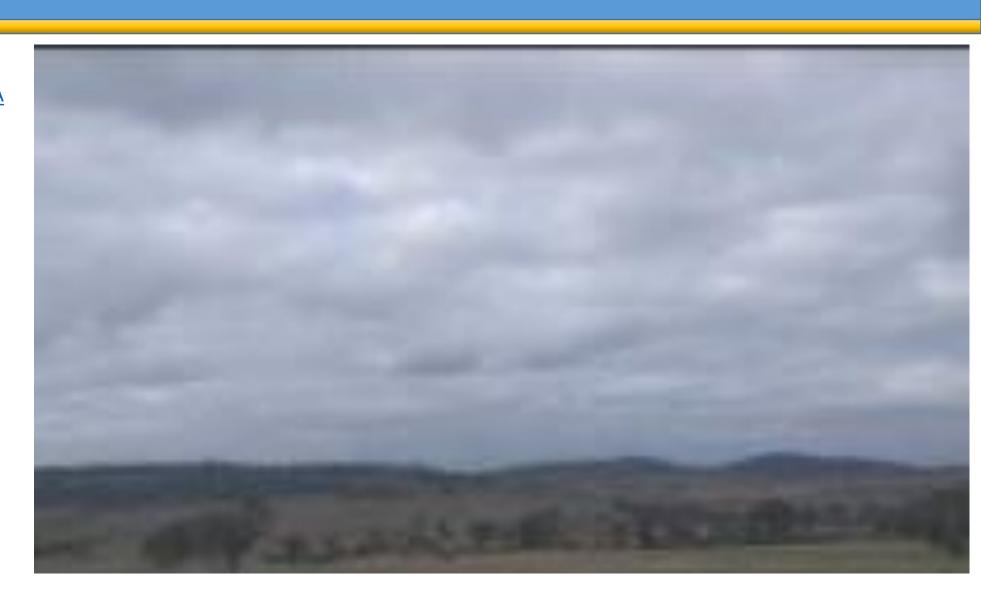
Marco Robustini's Explora https://youtu.be/hy-UXrUSplU



Questions?



Tridge's SwanK1 https://youtu.be/iRpVvW-AAyA



Questions?



https://youtu.be/ktcWGpBBqEQ

